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31	January 1972	

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If further information is required, please let us know.

Regards,	
Project Engineer	
Advanced Development Pr	coiects

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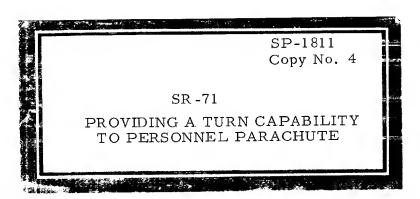
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# Lockheed Aircraft Corporation

## ADVANCED DEVELOPMENT PROJECTS

REPORT NO.	ξ_	SP-1811
DATE	<u>{</u>	11 Nov. 1971
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MODEL

SR-71

TITLE

PROVIDING A TURN CAPABILITY TO PERSONNEL PARACHUTE

PREPARED BY

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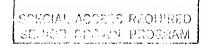
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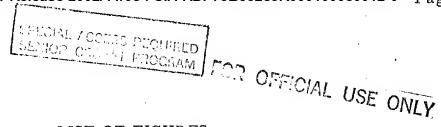
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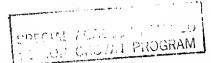
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### LIST OF FIGURES

FIGURE 1-1	RISER LINES
FIGURE 1-2	RELEASE LINE
FIGURE 1-3	TEST JUMPER DESCENT - 3 LINES RELEASED
FIGURE 1-4	TEST JUMPER DESCENT - 6 LINES RELEASED
FIGURE 1-5	PARACHUTE

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### ABSTRACT

This report describes the modification to, and the tests of, the air crew parachutes, the objective of which was to provide a crew initiated and controlled turn (steering) capability during a descent.

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#### **INTRODUCTION**

The personnel parachutes used on the SR-71 are not standard USAF personnel parachutes. The standard chutes are smaller and flatter and are capable of being turned during descent. The turning technique requires the release of four of the suspension lines from the canopy to the risers attached to the crewman's harness. Following this, the crewman, by chinning himself on one of the four risers, can cause the canopy to turn and thus can control his heading and, therefore, to some extent control his landing point to avoid obstructions.

This report describes the attempt to provide a similar feature to the larger diameter and extended skirt canopy parachute used by the SR-71 air crew.

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The intent of providing turn capability to the SR-71 parachute is to increase an air crewman's chances for survival.

However, fast reaction and high physical strength are required for successful use of the six line release method on the SR-71 parachute system. These requirements could not be met even with our extremely rigorous training programs. Also incorporating this technique will increase the complexity of parachute packing.

Therefore, it is concluded that this technique actually detracts from an air crewman's ability to survive and it is not recommended that it be incorporated in the SR-71 parachute system.



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#### DESCRIPTION OF MODIFICATION

An examination of Figure 1-5 will disclose the important anatomical features of the SR-71 air crew parachute when fully deployed. Following tests described in other sections of this report, it was determined that the optimum number of suspension lines to be released was six; three lines from each aft riser to the canopy.

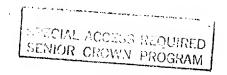
Figure 1-1 gives an overall close-up of the risers and the modification to the aft risers to incorporate release lines to effect the release of the three suspension lines mentioned above.

Figure 1-2 is a more detailed close-up of the upper portion of one of the release lines and its attachment to the "D" ring to which the suspension lines from the canopy are attached.

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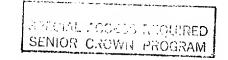
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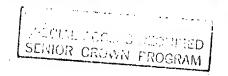
The purpose of the static tests was to determine the static strength of the suspension lines with and without the modification of adding the release line shown in Figure 1-2.

As shown in Figure 1-2 three of the eight suspension lines to the "D" ring were modified for these tests. Also full length suspension lines were used.

An examination of the following table of test results shows that the modification did not degrade the strength of the system.

Sample No.	$\underline{ ext{Type}}$	Failure Load Lbs.
. 1	Standard	3200
2	Modified	3250
3	Modified	3500
4	Modified	3000
5	Standard	3250
6	Standard	3250
7	Standard	3250
8	Standard	3250
9	Standard	3250
10	Modified	3500
,11	Modified	3000





#### WHIRL TOWER TESTS

The purpose of the whirl tower tests was (1) to test the pull force required on the release line to release the three suspension lines under simulated dynamic conditions and (2) to verify the structural integrity of the parachute lines and risers under simulated dynamic conditions.

(1) Three tests using 300 lb. dummies were run at 170 KIAS

to check pull forces required to release the three suspension
lines. The results were as tabulated:

Test No.	Pull For	ce Libs.
1	L. H. 40	R.H. 50
2	L. H. 45	R.H. 35
3	L. H. 60	R.H. 60

(2) A series of tests were run at various KIAS to verify structural integrity. The three tests at 170 KIAS reported above demonstrated the structural adequacy of the system at that KIAS.

Three tests were run at 300 KIAS with 300 lb. dummy and with a Navy test harness. In the first test, the "D" ring on the harness pulled out when the chute was approximately 3/4 open. In the second test, the neck ring on the dummy came away when the chute was approximately 3/4 open. In the third test there was no camera coverage and therefore no record of the sequence of failure. In all three of these tests the dummy separated from the chute. The release line modification was

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unaffected in all three tests.

An additional two tests were run at 225 KIAS and 250 KIAS respectively with complete success.

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#### AIRPLANE DUMMY DROP TESTS

The purpose for making drop tests with dummies from airplanes was to discover the optimum number of suspension lines to be released to render the parachute capable of being turned (steered) by the air crewman.

These tests were performed with a 300 lb. dummy being dropped from a C-130 at 10,000 feet altitude and at 110 - 125 KIAS.

The SR-71 parachute has seven (7) suspension lines on each of the two rear risers (see Figure 1-2) and eight (8) suspension lines on each of the two front risers.

The first drop tests determined the maximum number of suspension lines that could be released on the rear risers before the canopy collapsed. The canopy deployed stably with a total of eight lines released (four on each side). However, when 10 lines were released (five on each side) the canopy would alternately fill and collapse at approximately 50 foot intervals. This established that the maximum number of lines released per rear riser should not exceed three (3) per side in order to provide the required safety for live jumps.

It was further determined that for proper stability in terms of canopy action, that no more than three lines per riser should be released. Therefore, the optimum number of lines to be released to effect turn control of the chute during descent should be three per rear riser for a total of six.

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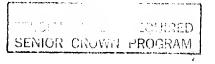
LIVE JUMP TESTS

The purpose of these tests was to evaluate the feasibility of the air crewman turning (steering) the parachute during descent.

The live jumps were made from a C-130 flying at 10,000 feet altitude and 110 - 125 KIAS. The first nine jumps were with "shirt sleeve" equipment to acquaint the men with the parachutes and other SR-71 equipment and the suspension line release. The other nine jumps were made with the men wearing full pressure suits. The technique of suspension line release used was to release three of the seven lines on each of the rear risers in consecutive order after the canopy opened (Figures 1-3, 1-4 and 1-5).

To summarize the written statements of the jumpers which are duplicated in the appendix:

The jumpers reported that it required approximately 30 seconds to make  $360^{\circ}$  turns with the six suspension lines released. Holding down a front riser to effect the turn was exhausting effort. The descent rate was unaffected by the release of the six suspension lines, remaining approximately 20 feet per second. Also a constant turning factor was introduced after line release which demanded continuous physical effort to overcome. The down wind velocity was increased by four knots at a ground wind velocity of eight knots.

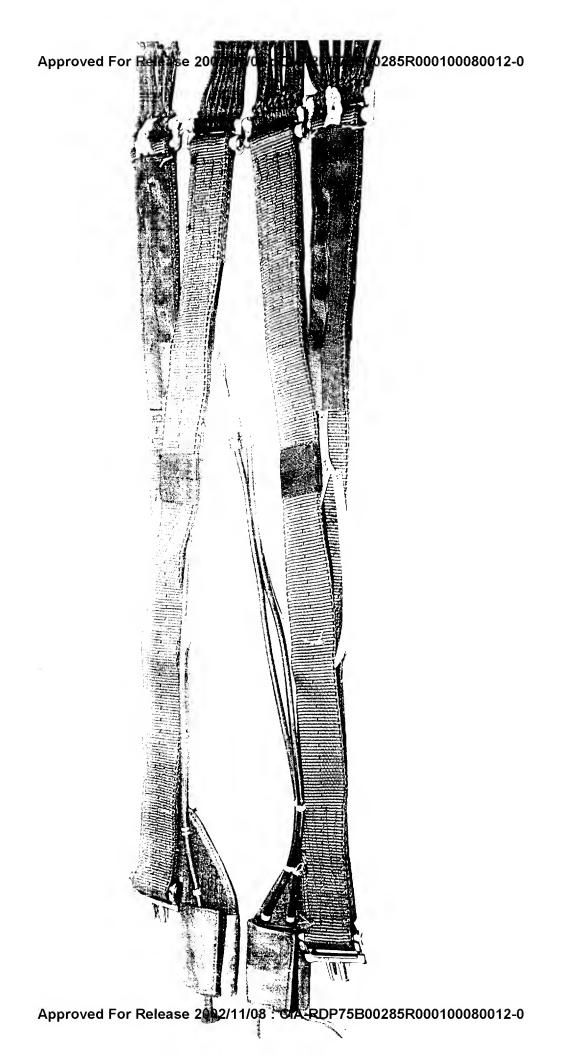




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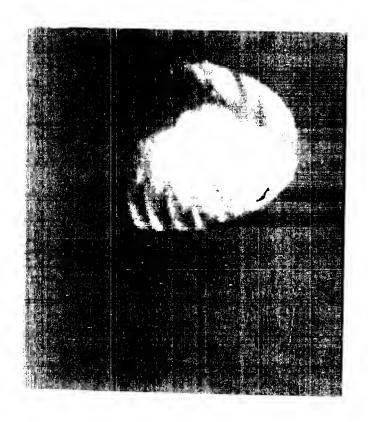
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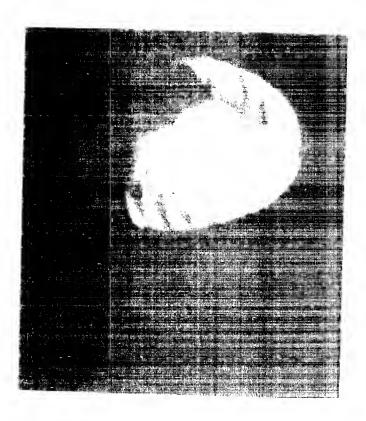
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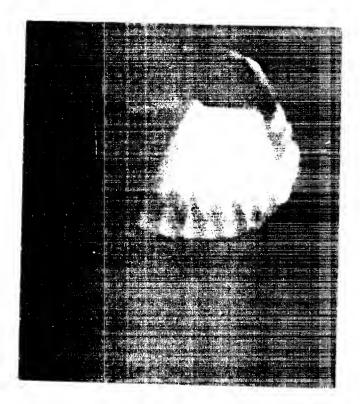
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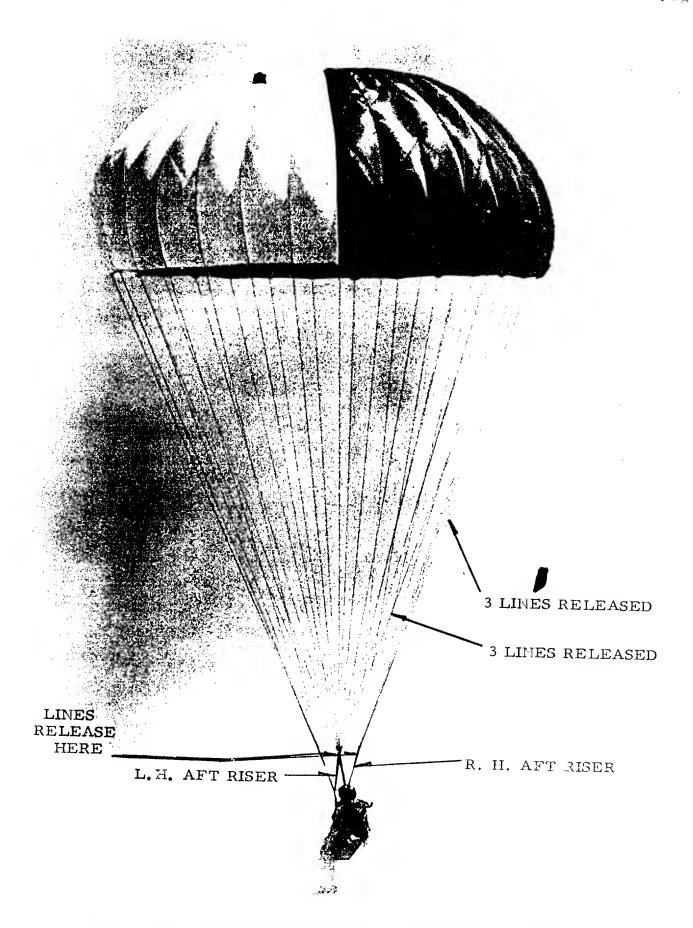
TEST JUMPER DESCENT

10 IMES OF AFT RISER RELEASED)





HIMPER DESCENT
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### APPENDIX

### Live Jump Test Reports

<u>Name</u>	Date
Simpkins, Jimmie C.	9 June 1971
	11 June 1971
	30 June 1971
	2 July 1971
Alexander, Guillermo M.	4 June 1971
	11 June 1971
	2 July 1971
	8 July 1971
Nye, James W.	9 June 1971
	11 June 1971
	8 July 1971
Sallee, Jettie L.	4 June 1971
	9 June 1971
	30 June 1971
	2 July 1971
Brown, Herbert R.	30 June 1971
•	8 July 1971
Powers, William E. Jr.	4 June 1971

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of the jumper so minute. The cano slips were much lappears to swing a	the left front i py made two 360 <sup>0</sup> ess'approximatal more in a front	riser was pu <sup>9</sup> turns to t Ly one 360 <sup>0</sup> -to rear mot	illed to chest the left. Tur turn in 80 so Ton va turnii	t level rna usi econda.	for appr. .ng rear o 	oximatel ne riser t kit ra	y 1 : ift	
of the jumper so minute. The cano slips were much lappears to swing a	the left front i py made two 360 <sup>0</sup> ess'approximatal more in a front	riser was pu <sup>9</sup> turns to t Ly one 360 <sup>0</sup> -to rear mot	illed to chest the left. Tur turn in 80 so Ton va turnii	t level rna usi econda.	for appr. .ng rear o 	oximatel ne riser t kit ra	y 1 : ift	
of the jumper so minute. The cano slips were much lappears to swing a	the left front i py made two 360 <sup>0</sup> ess'approximatal more in a front	riser was pu <sup>9</sup> turns to t Ly one 360 <sup>0</sup> -to rear mot	illed to chest the left. Tur turn in 80 so Ton va turnii	t level rna usi econda.	for appr. .ng rear o 	oximatel ne riser t kit ra	y 1 : ift	
of the jumper so minute. The cano slips were much lappears to swing a	the left front i py made two 360 <sup>0</sup> ess'approximatal more in a front	riser was pu <sup>9</sup> turns to t Ly one 360 <sup>0</sup> -to rear mot	illed to chest the left. Tur turn in 80 so Ton va turnii	t level rna usi econda.	for appr. .ng rear o 	oximatel ne riser t kit ra	y 1 : ift	
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of the jumper so minute. The cano slips were much lappears to swing a	the left front i py made two 360 <sup>0</sup> ess'approximatal more in a front	riser was pu <sup>9</sup> turns to t Ly one 360 <sup>0</sup> -to rear mot	illed to chest the left. Tur turn in 80 so Ton va turnii	t level rna usi econda.	for appr. .ng rear o 	oximatel ne riser t kit ra	y 1 : ift	
of the jumper so minute. The cano slips were much lappears to swing a	the left front i py made two 360 <sup>0</sup> ess'approximatal more in a front	riser was puo turns to to ly one 360° to rear motors ery minor on	illed to chest the left. Tur turn in 80 so Ton va turnii	t level rus usi econds. ng to s	for appr ng rear o The sea right or	oximatel ne riser t kit ra	y 1 : ift	

AFFTC FORM L-0-16

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	PMONATURE SERVERSE	MODEF19661: CIA-RDP75B00285尺0000000000129で1
AST NAME . FIRST NAME	· MIDDLE INITIAL	GRADE BERIAL NUMBER STAT
SIMPKINS, JIMMIE	c.	CPT
PANACHUTE TYPE		MAIN CANOPY TYPE
X BACK	SEAT	28' STANDARD FLAT SO' PERSONNEL T-10
CHEST	TROOP	24' STANDARD FLAT FOR OFFICIAL USE ONLY
ARACHUTE PART NR		MAIN CANOPY DATA
YES	□X NO	DAMAGE TWIST IN LINES
RESERVE PAR	RACHUTE DATA	TX NONE MEDIUM YES TX NO
JSED		LIGHT HEAVY NR OF FULL TWISTS None
YES	□Х но	SEMI-INVERSION COMPLETE INVERSION
UNUSUAL OCCURRENCE		YES X NO YES X NO
YE\$	СМ ко	SQUIDDING CANOPY INJURIES  YES X NO YES X NO
EMARKS I	.,	titude 10,000 - Delay - 3 sec. The jumper exited the
provided for pu	illing the risers	shoulder level. Average turn time was 35 sec. The tabs down were approximately 6" too high. The seat being emed to increase oscillation. The seat kit was release to the s
provided for pu with the jumper ed at 4,000 ft. approximately 9 comfortable and	lling the risers at this time se No problems.  O due to the ju I remained cool a	down were approximately or too night the seat being
provided for pu with the jumper ed at 4,000 ft. approximately 9 comfortable and	lling the risers at this time se No problems.  O due to the just remained cool a	down were approximately 6" too high. The seat being emed to increase oscillation. The seat kit was releas Turns attempted from this time to ground impact were mper being exhausted. The pressure suit is very lmost to impact. During the 360° turns the canopy
provided for pu with the jumper ed at 4,000 ft. approximately 9 comfortable and	lling the risers at this time se No problems.  O due to the just remained cool a	down were approximately of too high. The seat being emed to increase oscillation. The seat kit was releas Turns attempted from this time to ground impact were mper being exhausted. The pressure suit is very lmost to impact. During the 360° turns the canopy
provided for pu with the jumper ed at 4,000 ft. approximately 9 comfortable and	lling the risers at this time se No problems.  O due to the just remained cool a	down were approximately of too high. The seat being emed to increase oscillation. The seat kit was releas Turns attempted from this time to ground impact were mper being exhausted. The pressure suit is very lmost to impact. During the 360° turns the canopy
provided for pu with the jumper ed at 4,000 ft. approximately 9 comfortable and	lling the risers at this time se No problems.  O due to the just remained cool a	down were approximately 6" too high. The seat being emed to increase oscillation. The seat kit was release Turns attempted from this time to ground impact were mper being exhausted. The pressure suit is very lmost to impact. During the 360° turns the canopy
provided for pu with the jumper ed at 4,000 ft. approximately 9 comfortable and	lling the risers at this time se No problems.  O due to the just remained cool a	down were approximately 6" too high. The seat being emed to increase oscillation. The seat kit was release Turns attempted from this time to ground impact were mper being exhausted. The pressure suit is very lmost to impact. During the 360° turns the canopy
provided for pu with the jumper ed at 4,000 ft. approximately 9 comfortable and	lling the risers at this time se No problems.  O due to the just remained cool a	down were approximately of too high. The seat being emed to increase oscillation. The seat kit was releas Turns attempted from this time to ground impact were mper being exhausted. The pressure suit is very lmost to impact. During the 360° turns the canopy
provided for pu with the jumper ed at 4,000 ft. approximately 9 comfortable and	lling the risers at this time se No problems.  O due to the just remained cool a	down were approximately of too high. The seat being emed to increase oscillation. The seat kit was releas Turns attempted from this time to ground impact were mper being exhausted. The pressure suit is very lmost to impact. During the 360° turns the canopy
provided for pu with the jumper ed at 4,000 ft. approximately 9 comfortable and	lling the risers at this time se No problems.  O due to the just remained cool a	down were approximately of too high. The seat being emed to increase oscillation. The seat kit was releas Turns attempted from this time to ground impact were mper being exhausted. The pressure suit is very lmost to impact. During the 360° turns the canopy
provided for pu with the jumper ed at 4,000 ft. approximately 9 comfortable and seemed to turn	lling the risers at this time se No problems.  O due to the just remained cool a	down were approximately of too high. The seat being emed to increase oscillation. The seat kit was releas Turns attempted from this time to ground impact were mper being exhausted. The pressure suit is very lmost to impact. During the 360° turns the canopy
provided for pu with the jumper ed at 4,000 ft. approximately 9 comfortable and	lling the risers at this time se No problems.  O due to the just remained cool a	down were approximately 6" too high. The seat being emed to increase oscillation. The seat kit was releas Turns attempted from this time to ground impact were mper being exhausted. The pressure suit is very Impost to impact. During the 360° turns the canopy 45° jerking movements.
provided for pu with the jumper ed at 4,000 ft. approximately 9 comfortable and seemed to turn	lling the risers at this time se No problems.  O due to the just remained cool a	down were approximately 6" too high. The seat being emed to increase oscillation. The seat kit was releas Turns attempted from this time to ground impact were mper being exhausted. The pressure suit is very Imost to impact. During the 360° turns the canopy 45° jerking movements.
provided for pu with the jumper ed at 4,000 ft. approximately 9 comfortable and seemed to turn	lling the risers at this time se No problems.  O due to the just remained cool a	down were approximately 6" too high. The seat being emed to increase oscillation. The seat kit was release Turns attempted from this time to ground impact were imper being exhausted. The pressure suit is very important to impact. During the 360° turns the canopy of 45° jerking movements.
provided for pu with the jumper ed at 4,000 ft. approximately 9 comfortable and seemed to turn	lling the risers at this time se No problems.  O due to the just remained cool a	down were approximately 6" too high. The seat being emed to increase oscillation. The seat kit was release Turns attempted from this time to ground impact were imper being exhausted. The pressure suit is very limost to impact. During the 360° turns the canopy of 45° jerking movements.  Special Coess Required program

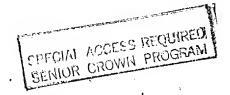
etc. NEGATIVE REPORT IS REQUIRED.

I This space to be used to explain all, unusual occurrences, injuries,

JIMMIE C. SIMPKINS, CPT, USA

				Page 19
	INDIVIDUAL LIVE J	UMP REPORT	July 1971	-
SIMPKINS, JIM		S CORE ACCESS REC SENIOR CROWN PRO	OGRAM	STA
PARACHUTE TYPE		MAIN CANOPY TYPE	LISE ONL	
X BACK	SEAT	MAIN CANOPY TYPE  20. FTORAROFETACIAL	30, BERSONNET	
CHEST	TROOP		манто 🔀	
PARACHUTE PART NR				
UNUSUAL OCCURRENCE			CANOPY DATA	
YES	X NO	DAMAGE	TWIST IN LINES	
RESERVE PA	ARACHUTE DATA	X NONE MEDIUM	YES X	но
USED	•	LIGHT HEAVY	NR OF FULL T	WI5T5
T YES	⊠ ио	SEMI-INVERSION	COMPLETE INVERSION	
UNUSUAL OCCURRENCE		YES X NO	YES X	) NO
		SQUIDDING CANDPY	INJURIES	
T YES	. 🔼 ио	TYES X NO	YES X	) NO
BEMARKS I				

Project - JON 134ADO - Altitude 10,000 ft - Speed 125 KIAS. The jumper exited the aircraft in an airborne position and delayed for 3-5 seconds. The ripcord pull was easy; however, the ripcord was hard to see from the helmet. The jumper released the 6 lines with no problems. 360° turns were accomplished by use of the left front riser. Approximate time was 30 seconds per 360°. The jumper around 6000 ft set a heading for the bull's eye on the drop zone and was able by pulling on the right front riser and then the left front to hold the heading for approximately 2000 ft. The seat kit was released at 4000 ft. No problems. Turns attempted with the canopy after seat release were minimum due to seat kit revolutions below the jumper and also the jumper's strength was exhausted. The pressure suit was comfortable. No problems were encountered in this test.



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oic. NEGATIVE REPORT IS REQUIRED.

JIMMIE C. SIMPKINS, CPT, USA

		Page 20
Approved Balance lease &	02FPI708 : CIA-RDP75B00285D0	001,000,800,92.0 = /9//
LAST NAME . FIRST NAME - MIDDLE INITIAL	GRADE GRADE	SERIAL NUMBER
	1.ST / TOUR	
ALEXANDER GUILLERMO	M ST 17 CUIB	
PARACHUTE TYPE	MAIN CANOPY TYPE	
	28'STANDARD FLAT	SOV PERSONNEL TOTO
BACK DEAT	TISE OF	
CHEST TROOP	FOR OF AND ABOUT SE	OTHER STAT
PARACHUTE PART NR		A ·
UNUSUAL OCCURRENCE		OPY DATA '
☐ YES NO	DAMAGE	
RESERVE PARACHUTE DATA	NONE MEDIUM	L YES - NO /
U•ED .	LIGHT HEAVY	NR OF FULL TWISTS
T YES NO	SEMI-INVERSION	COMPLETE INVENTION
UNUSUAL OCCURRENCE	TYES NO	YES NO
	SQUIDDING CANOPY	INJURIES
□ YE\$ NO	YES NO	YES NO
REMARKS I		, ,
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land deployed	he constant	$O_{II}$
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To the act a	Vacture. Du	Jas tired when or
the risers are tens for the conector le	Miles .	
I This space to be used to explain all unaual occur	rences, injuries, A. M.	mi Mana
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Approx For B	Eleselzeos/ablost C	IA-RDP75B00285F	000100080012-07/	age 21
LAST NAME . FIRST NAME . MIDDLE INITIA	(L	RADE	SERIAL NUMBER	
ALEXNUER, GUILLEINE	m	1St 1T		
BACK SEAT	MAIN CANOPY	· ·	AL ACCUMAN DIO	190 V
CHEST TROO		ANDARD FLAT	F A S 1 S 1 S 1 S 1 S 2 S 3 S 3 S 3 S 3 S 3 S 3 S 3 S 3 S 3	STAT
PARACHUTE PART NR			ANOPY DATA	
UNUSUAL OCCURRENCE	DAMAGE		TWIST IN LINES	<del></del>
RESERVE PARACHUTE DATA	NONE	MEDIUM	TYES THE	
USED			NR OF FULL TWISTS	
	LIGHT			
☐ YES [V] NO	SEMI-INVERSIO		COMPLETE INVENSION	
UNUSUAL OCCURRENCE	YE5	III NO	☐ YES ☑ NO	
☐ YE\$ NO	SQUIDDING C	NOBA LA	THIURIES NO	
REMARKS I				
Exit the amplon		•		
Pulled. After out	AND OPEN :	I RELEASEP	THE bLINES AND	<b>)</b>
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TAKES A LOT OUT	of you.	J		)  ) 
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		-1/08 :- GIA-RDP75B00286F	SP-1811
	INDIVIDUAL LIVE JUM	P REPERT	Page 23 とてひん 7/ STA
ST NAME . FIRST NAME	- MIDDLE INITIAL	GRADE	SERIAL NUMBER
ALEXANDER PARACHUTE TYPE	E GUILLEPA	MAIN CANOPY TYPE	
BACK CHEST	TROOP	28° STANDARD FLAT	SPECIAL ACCESS REQUIRED
PARACHUTE PART NR UNUSUAL OCCURRENCE		м	AIN CANOPY DATA
YES .	Muo	DAMAGE	TWIST IN LINES
RESERVE PAR	RACHUTE DATA	NONE MEDII	
YES	≥ NO	SEMI-INVERSION	COMPLETE INVERSION
UNUSUAL OCCURRENCE		SQUICOING CANOPY	INJUNIES NO
YES	. но	☐ YES NO	□ YES NO
was 10,00 about 2, after che lines a turns. The it was to	soft. Ap sec and a redoing to not down	clic I left deployed the lie canopy of in desent of una d music E reser. To	the C-130 Took the C-130 Thook was a lot lessed was
works to	make tu	rns without	The surrival hi
		SENION	ACCESS REQUIRED CROW 1 PROGRAM  CIAL USE ONLY
1 This app Approy	ed.For,Release.2002/1	1/08 : GIA-RDP 75 6 0 285 R	300010008Ø012-0
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	APPROVED DE LE	925912002/21 <del>108</del> 1	CIA-RDP75B002	852000100080012-	0, 7/
LAST NAME FIRST	NAME . MIDDLE INITIAL	1	GRADE ,	SERIAL NUMBER	
Aye-	Sames	w	Capi	300 1 - 00	
PARACHUTE TYPE		MAIN CAN	PY TYPE	The state of the second second seconds	STATA e
BACK	SEAT	28	STANDARD FLAT	50' PERSONNEL	€ T-19
CHEST	TROOP	24	STANDARD FLAT	R OFFICIAL U	se only i
PARACHUTE PART I				AIN CANOPY DATA	
YES	₩ но	DAMAGE		TWIST IN LINES	
	VE PARACHUTE DATA		NE MEDIU	M YES	□ но
USED				1	ULL TWISTS
	□ NO	יים נים			
☐ YE#	□ . но	SEMI-INVE		COMPLETE INV	
UNUSUAL OCCURRE	NCE			YK#	<u> Р</u> но
	мо 💢	SQUIDDING		INJUNIES	
T YES	مربر سو م	YE	:s' X NO	☐ AK\$	Ø_NO
1. NO RELEAS	RISA TURNS. X OFLINES, NO KIT	neceuse. Taide	ED PULLING A YIS	cer 30sec, No To	RNING.
3. KIT RECEN	ces, 3 line released, 6 line relations	e. Effect so	60° turn in	- // 1	•
eft ext	nely tiring	pulling ris	opy here	mend pilo	
			-	CHULL PROGRAM	· ·
1 This space	to be used to explain all us	nueval occurrences, inju	SIGNATURE		

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etc. NEGATIVE REPORT IS REQUIRED.

	Approxydi500 Belevee 800	244408, CIA-RDP75B00285B00	0100080012-0 Page 25
LAST NAME . FIRST	NAME - MIDDLE INITIAL	GRADE	SERIAL NUMBER
Nyc	James W	Carpt CIAL ACI	
PARACHUTE TYPE		MAIN CANOPY TYPE	VIN PROGRAM STAT
BACK	SEAT	28' STANDARD FLAT	SO PERSONNEL THO
CHEST PARACHUTE PART I	TROOP	A STANDARD PEOR OF	TOTAL USE ONLY
UNUBUAL OCCUMBE			NOPY DATA
	NO NO	DAMAGE	TWIST IN LINES
USED	/E PARACHUTE DATA	NONE MEDIUM	☐ YES ☐ NO
☐ Y E #	□ NO	SEMENVENTION HEAVY	NR OF FULL TWISTS
UNUSUAL OCCURRE	NC E	[T] YES [T] NO	COMPLETE INVENSION
¥#.	(T) NO	SQUIDDING CANOPY NO	INJURIES NO
CONNECTOR	LINKS PULLED	TO CHIN LEVEL	TO INITIATE TUENS
ONLY &	fective TURNS	were mocomplished	Lusing front
risers.	UNSTABLE ON OFF	NING Some Damag	e to deoxple of
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		by polling on a fre	
		tof the trans. Ma	
,		recommend a pelo	
he is 200	300 ft be tore	nating turns. Ats	1 Short
		Jumps, under left ou	
•		SPECIAL / SENIOR C	OCEAS REQUIRED PROGRAM
•			
	·	FOR OFFICIAL US	SE ONLY
1 This scare to h	Se used to explain all served assert	SIGNATURE	
	> used to explain all uniaual occurrer 'C REPORT IS REQUIRED.	rece, injuries,	Nyc

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		Marie Contract	DI 1000020	₹000100080012	SP-18
	INDIVIDUAL LIVE JU	MP REPORT		8	-Juny 71
ST NAME - FIRST NAM	E . MIDDLE INITIAL	G	SENIOR	ACCE MENTAL NU	MBER
NIE JA	rmes W		CIPI	CROW	
ARACHUTE TYPE	. '	MAIN CANOPY	TYPE		STAT
BACK	SEAT	[] 25° 5T	ANDARD FLAT	30' PERSON	
CHEST	TROOP	24' STA	NDARD FLAT	OTHER	. The Colombia
RACHUTE PART NR			N	AIN CANOPY DATA	
YES .	Ø NO	DAMAGE		TWIST IN L	INES
	ARACHUTE DATA	MONE	MEDI		7
SED.		LIGHT	· EA	Y NR	OF FULL TWISTS
YES NUSUAL OCCURRENCE	, no	SEMI-INVERSIO	и М	COMPLETE YE	E INVERSION
		SQUIDDING CA	•	INJURIES	
T YES	PI NO	YE\$	□ K NO		8 D NO
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As a res axdmad are pull The pres difficul	sult it is more tir does not have a te led. That is it ho sure suit did not lt to see because o	ing making to to dency to to local to l	turns. The arm unless to well. To colem. It, but the contract the cont	canopy kakda the front rise was a little	ers more

1 This specApproved For Release 2002/11/08: CTA-RDP 75B00285R000100080012-0

	PARONOAPENOENA	2002/11/08r. CIA-RE	P75B00285	R0001000800 <del>12-</del> 0	Page 27
	The state of the s	GRADE		SERIAL NUMBER	<u> </u>
LAST NAME . FIRST NAME		SENIOR CRIMINT	ROGRAM		
- 10/100,	and Printer of the same of	DR. REELSVAVA	The second of th		
PARACHUTE TYPE	٠	MAIN CANOPY TYPE		ST	AT
BACK	SEAT	28' BTANDAR		30. BEURONNET	T-10
CHEST	TROOP	24° STANDAR	D FLAT	OTHER.	
PARACHUTE PART NR UNUSUAL OCCURRENCE			MAIN	CANOPY DATA	
YES	<b>≅</b> 110	DAMAGE		TWIST IN LINES	
RESERVE I	PARACHUTE DATA	HONE	MEDIUM	YKS	LL TWISTS 3
	Jano No	LIGHT	HEAVY	COMPLETE INVEN	
UNUSUAL OCCURRENCE		SEMI-INVERSION YES	<u></u> но		~□′ но
·		SQUIDDING CANDRY		INJURIES	
	₩0	T YES		☐ YK.5	NO NO
REMARKS 1	le rear r	isers how	و معرف ما المعرب	HIO TU	rwing
ł	, Pulling				
will	produce	a 180°	Turn	in app	rux. 15=18
secono	G. Pullin	g one rea	or ris	ser wil	11 produce
a 5/i	ght Turn	but wi	11 res	ult in	Total
Exhous	tion it	held lo	ng tr	roughito	produce
	w. The			•	
Slight	Forward	drift.	Cappio	y. A KNO	ts), Sandi
was No	ormal.	The Ri	sers a	are hard	to pull
down	and hol	d. No	Oscil	lations u	vere
Observ			SPECIAL SENIOR	MCCESS REQUIRER CROWN PROGRAM	
•	o used to explain all unusual oc E REPORT IS REQUIRED.		IIGNATURE		2 112

Approvadition Pelogissus	02(£1/02 T CIA-RDP75B00286D00	Page 28 0100080012-0
LAST NAME . FIRST NAME . MIDDLE INITIAL	GRADE	SERIAL NUMBER
50 00, Jotio 1.	- COSS I	
PARACHUTE TYPE	MAIN CANORY TYPE CIAL ACCESS TO THE CONTROL OF CROWN  28' STANDARD ELAT  24' STANDARD RATOFFICIAL  24' STANDARD RATOFFICIAL	STAT ,
BACK BEAT	28' STANDARD ELAT	So. BERRONHER
CHEST TROOP	24' STANDLED RATOFFICE	OTHER
PARACHUTE PART NR		$r_{ij}$
UNUSUAL OCCURRENCE	, MAIN CA	NUFTUNIA
YES NO	DAMAGE	TWIST IN LINES
RESERVE PARACHUTE DATA	NONE MEDIUM	YES (D'. NO
	LIGHT HEAVY	NR OF FULL TWISTS
☐ YEs S☑ NO .	SEMI-INVERSION	COMPLETE INVERSION
UNUSUAL OCCURRENCE	YES Y NO	☐ YES ,☐, NO
☐ YES NO	SQUIDDING CANOPY	☐ AR2 ☐ NO
Remarks Par C-1	30 - 110 Knots	- 10,000 flat
Normal apening	shock . Relia	sed fit and
oliserved. Kellease		Bo. Turning
stady pull. Re was real slow.	Released The Released	se worked ok;
swing lke starte	skausing. Using	g Ile front
	- A - A	
in approps . 25-30.	sec. Hear Forward	Driff is approx.
3-4 Knots. Tank	ing was mound	SPECIAL ACCESS REQUIRED SENIOR CROWN PROGRAM
	EOI	R OFFICIAL USE CIALY
	SIGNATURE .	
1 This space to be used to explain all unusual occur		

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2		DATE Page 29
Appknykoldingselogssund	02/64/087 CIA-RDP75B00285B00	0100080012-0 ~/
LAST NAME - FIRST NAME - MIDDLE INITIAL	SENIOR CAPEVIL PROGRAM	SERIAL NUMBER
Sallee Tettie K	SEFATUSE C	
PARACHUTE TYPE	MAIN EANOPYTYPE	STAT
BACK BEAT	28° STANDARD FLAT	30' PERSONNEL T-10
CHEST TROOP	24' STANDARD FLAT	OTHER AND
PARACHUTE PART NR UNUBUAL OCCURRENCE	MAIN CAN	OPY DATA
☐ YES ☑ NO	DAMAGE	TWIST IN LINES
RESERVE PARACHUTE DATA	NONE MEDIUM	X YES D NO
USED	LIGHT . HEAVY	NR OF FULL TWISTS
☐ YES ☑ NO	SEMI-INVERSION	COMPLEYE INVERSION
UNUSUAL OCCURRENCE	☐ YES ☑ NO	☐ YES ☑ NO
☐ YES ☑ NO	SQUIDDING CANDRY	AES NO
REMARKS		A
Ramp Exit C-130	125 KNOTS, 10,0	oo fit.
	-	
Normal Opening.	Small Tear	in rear of
worme. options	· · · · · · · · · · · · · · · · · · ·	11: Right Front
Canopyo Released riserndown to eye	6 lines. Pu	illng willing
ONNector	lovel and Hold	ing produced
riser A down To Ele	,	o Ilimo Pett
•	a of some	$-u \cap v_{\mathcal{F}}$
riser connector dou	in to eye Level	produced a
riser connector dou		
360° Turn in app	10x 30 sec 01	v one Turn
360° Turn IN app		
To the Left T	he Turn come	in steps of
		4 /
Approx. 300 This	was only not	riced on
one Turn. The	e conopy was	constantly
Trying To turn	Peft. Relea	sed Seat
Kit at 4000 ft.	Turning Abi	lity is Decreased
with Kit released	· \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \	THE PROGRAM OVER)
I This space to be used to explain all unusual occurs etc. NEGATIVE REPORT IS REQUIRED.	rences, injuriee,	

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# Approved For Delease 2002/11/08: CIA-RDP75B00286R000100080012-0

extremely Difficult. with The Kit and Roft. Swinging Counter-Clockwise a Left Turn can be made in approx. 30 sec.

harder than with no pressure suit is slightly is still Comfortable.

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EOR OFFICIAL USE ONLY

BACK SEAT S8' STANDARD FLAT S0' PERSONNEL  CHEST TROOP 24' STANDARD FLAT OTHER  PARACHUTE PART NR  UNUSUAL OCCURRENCE MAIN CANOPY DATA  TYES NO DAMAGE TWIST IN LINES  RESERVE PARACHUTE DATA	SP-1811 - 1/age 31 - 1/
LAST NAME - FIRST NAME - MIOOLE INITIAL    SOURCE   SERIAL NUMBE	$\sim$ $\sim$ $\sim$ $\sim$
MAIN CANOPY TYPE  BACK BEAT 28' STANDARD FLAT 30' PERSONNEL  CHEST TROOP 24' STANDARD FLAT OTHER  PARACHUTE PART NR  UNUSUAL OCCURRENCE MAIN CANOPY DATA  TYES NO DAMAGE TWIST IN LINES  RESERVE PARACHUTE DATA	
BACK SEAT S0' PERSONNEL  CHEST TROOP 24' STANDARD FLAT OTHER  PARACHUTE PART NR  UNUSUAL OCCURRENCE MAIN CANOPY DATA  TYES NO DAMAGE TWIST IN LINES  RESERVE PARACHUTE DATA	~~ ~ ~
PARACHUTE PART NR  UNUSUAL OCCURRENCE  YES  NO  DAMAGE  TWIST IN LINES	STAT
TWIST IN LINES  RESERVE PARACHUTE DATA  MAIN CANOPY DATA  TWIST IN LINES	
RESERVE PARACHUTE DATA	
NONE MEOLUM YES	
LIGHT HEAVY NR OF 1	FULL TWISTS
UNUSUAL OCCURMENCE SEMI-INVERSION COMPLETE INV	ERSION NO
YES YES YES YES	
Steped off ramp and went unstable.  Shock was extremely hight.  Made a 6 fine Release and Time Several 3600 Turns. The 360° Turn	ned swere
fastest sturm was to the left which completed in just under 25 sec.	ne
AT. 4000 ft. I attempted to release Ret. The Kit released on the ly only. I pushed on the ket twice	e to
get it & release. When the Rit re the Rit release. When the Rit re the droped free and fall to the greater FOR OFFICIAL GIVE FOR OFFICIAL GIVE REDUISED.  1 This space to A proved to British & 2002/11/08 TOTA-RDP 15B00285 ROOM 100080012-0	eleased of with

FORM

AFFTC

Oscillations seemed to be larger without

the Bit below.

OFFICIAL USE SPECIAL ACCESS REQUIRED SENIOR CROWN PROGRAM

Turning was quicker with no kit.

I had a small oscillation on landing

SPECIAL ACCESS REQUIRED SENIOR CROWN PROGRAM

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				one and Marketin Section of the last	25-18II	
	PROYOTAL PEPERUM			Z	July 19/1	
LAST NAME - FIRST NAM	AE - MIDDLE INITIAL	GR	ADE	SERIAL NUM		
Brown, Herbert R			1st Lt		STAT	
PARACHUTE TYPE		MAIN CANOPY	Loc	NIOR CROWN PI	ROME	
X BACK	= SEAT		NDARD FLAT	30, DEMRON	T-10	
CHEST TROOP PARACHUTE PART NR		TOR OFFICIAL 00- VIII				
UNUSUAL OCCUMRENCE	II KIT.	MAIN CANOPY DATA DAMAGE TWIST IN LINES				
I IT'S VEG	deploy No	DAMAGE		I HIST IN CIT		
RESERVE P	ARACHUTE DATA	XX NONE	MEDIU	M CYKS	XX NO	
UBED		LIGHT	HEAV	Y NR C	OF FULL TWISTS	
YES	ом ХХ					
UNUSUAL OCCURRENCE		SEMI-INVERSION	<u>K</u> X но	COMPLETE YES		
	ON KX	SQUIDDING CAN	үчон ХХ	INJURIES YES	<u>Х</u> Х но	
REMARKS !						
to jar the kit l from the parachu above ground I a enough and was 1	e right hand side relloose. At approximalite harness and held attempted to toss the forced to land on it jump or landing.	tely 1,000 the kit in e kit away	ft, I relea my hand. a from me. I	sed the kit a At approximat did not toss	ttaching hardware ely 10 to 20 ft the kit far	
			٠.		•	
			•	•		
			- 1			
n a			ſ			
		**************************************			٠.	
V. Care	· · · · · · · · · · · · · · · · · · ·	$\int_{0}^{\infty} \int_{0}^{\infty} dt$	$\mathcal{N}$			
				~ DE38 REQU	IRED GRANN	
			SPEC	OFFICIA	r use only	
			SIGNATURE !	-Off-LILL		
	used to explain all unusual occur REPORT IS REQUIRED.	rences, injurisa,		6.1/18		

Approve	ed For Release 2002/	/11/08 : CIA-RDP75B00285B600100080012-0			
		SP-1811			
ı	NDIAIDNYT FIAE INW	PREPARTI ACCESS REQUIRED Page 34  SENIOR CROWN PROGRAM 8 July 1971			
ST NAME - FIRST NAME -		FOR OFFICIAL USL SERIAL NUMBER  STATE  STATE			
BROWN, HERBERT R.		FOR UNIST Lt			
PARACHUTE TYPE		MAIN CANOPY TYPE			
X BACK	SEAT	28' STANDARD FLAT 30' PERSONNEL T-10			
CHEST	TROOP	☐ 24' STANDARD FLAT ☐ OTHER SR-71			
PARACHUTE PART NR					
UNUSUAL OCCURRENCE		MAIN CANOPY DATA			
YES	(Х) ио	DAMAGE TWIST IN LINES			
RESERVE PARACHUTE DATA		NONE MEDIUM YES NO			
USED	. •	LIGHT HEAVY NR OF FULL TWISTS			
YES	[X] 40	SEMI-INVERSION COMPLETE INVERSION			
UNUSUAL OCCURRENCE		YES X NO YES NO			
		SOUIDDING CANOPY INJURIES			

I exited the C-130 aircraft from the ramp in a stable position. The suspension lines were released just after the canopy achieved a stable full open condition. Several 360 degree turns were accomplished with a time duration varying from 20 to 35 seconds. I found the canopy easier to turn during this test (without survival kit) than during a previous test with an undeployed survival kit.

[∑] но

T YES

REMARKS

SQUIDDING CANOPY

X No

T YES

SCEOINL CONSUS REQUIRED SENIOR OR WIN PROGRAM

YES

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etc. NEGATIVE REPORT IS REQUIRED.

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LAST NAME . FIRST NAME . MIDDLE INITIAL	GRADE COFOLA	SERIAL NUMBER				
	SPECIAL ACC					
POURRS William	= Ja. 115-61 CM	<u> </u>				
PARACHUTE TYPE	MAIT CANOPY TYPE	STAT				
BACK SEAT		30' PERSONNEL T-10				
CHEST TROOP	= 24' STANDARD FEOR OFF	HOM USE ONLY				
PARACHUTE PART NR						
UNUSUAL OCCURRENCE		IOPY DATA				
YES NO	DAMAGE					
RESERVE PARACHUTE DATA	MEDIUM	YES G-NO				
	LIGHT HEAVY	NR OF FULL TWISTS				
YES Z NO	SEMI-INVERSION .	COMPLETE INVERSION				
UNUSUAL OCCURRENCE	SQUIDDING CANOPY	INJURIES NO				
YES NO	YES Z NO	TYES A- NO				
REMARKS						
C-130 - 110 KTS - RAMP exit 10,000 FT.						
I stoped off	the tomp back	cwards facing				
Line OF Flight And Fell stable For Apox						
3. soc. Then I pulled The reprod with my						
right have AND The paraclas doployed And						
opened with out and notically opening stack.						
AFTER Full opining I thied to ture The						
canoly by pulling down The right Fruit friends +150%.						
This cousal slight occillation and No conogy						
tard. Then I released the SIX live to home by						
pulling down on both IAYUNDER SIMPSTRINIONED Then						
I referred The sent kit by pulling The house of the right side. As the Kit Fill noway The house						
Storyed attached to	17. SO T LAIS	SPANTANCOUS TOWNED				
I This space to be used to explain all unusual occurrences.  * etc. NEGATIVE REPORT IS REQUIRED.	ences, Injuries, A), 200	J. A.				

AFFTC FORM Approved for Release 2002/11/28 CHARDRISE DIVINE ON THE FOR OFFICIAL USE ONLY

up Approved For Depass 2002/168: CA-PDP7/B06285D000160080012-0 Find Worked the house from, The SPECIAL CREATING KIT SENTOR CREATING KIT SENTOR CREATING KIT SENTOR CREATING KIT OF THE PARTY I INDICATE OF THE PARTY OF THE PARTY OF THE OFFICE AND CREATING CR

I than third to Turn the compay by pulling on the hight tops hist. After helving The pisot down to whom the connector link was even with my fore head for 30 soc I had made the About A 10° Turn. By this time my prome were tired And I teleased the pison the convey the convey the convey the transfer to the original heading. After resting for A few socons I third turnling. After resting the front histore if took 15-18 soc for A 180° turn its ing the last front histore.

DECOST AND IANding were normal.

I don't feel Any tomber benefit

CAN be derived by using the rear Assers For

thrushy. Even using the Front bisons a man

would be exthemally tived thying to much a

complete canopy two.

[Complete Canopy two.

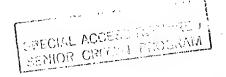
## Approved For Refease 2002/11/08 : CIA-RDP75B00285R900100080012-0

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### REFERENCES

SP-1650 Report, "SR-71 Personnel Parachute Descent Control"

T.O. 14D1-2-81, "Four Line Release"



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# Approved For Release 2002/11/08: CIA-RDP75B00285R000100080012-0

DETACHMENT 51 HQ SMAMA (AFLC)
NORTON AIR FORCE BASE, CALIFORNIA 92409

REPLY TO ATTN OF

5. 5. 61.3

ME

FEB 8 1972

Modified MC-1 Personnel Parachute

SAC(LGMR/Major J. Janis)

- 1. On 18 December 1969, an incident occurred during a test flight of an SR-71 aircraft which resulted in crew bail-out. The parachute descent ended in a near mishap to crew members on landing as indicated in the accident board investigation:
- a. "There was considerable oscillation after release of the seat kit...The landing was in a marginal area, in a rock pile, and I hit backwards and was immediately knocked back over onto my head and the chute fell over a cliff. My concern was that it was going to billow and drag me over the cliff and it appeared to be 150-200 feet down; however, the chute did not blossom...".
- b. "In the drogue chute, I was trying to keep my seat oriented toward the Pilot...Once I got the chute, it was very difficult to turn the chute to look at him. I could pull down the riser and rotate the chute to where I was looking at him; however, as soon as I would let go it would rotate back and I would be facing primarily in a southern direction... I hit approximately six feet from a ravine. My chute canopy went over the bank of the ravine. The chute collapsed immediately before I could activate the quick releases...had the canopy remained open and had there been any surface wind, I'm sure it would have pulled me into the ravine. The ravine was approximately 40 to 50 feet deep with fairly steep sides".
- 2. Because of the problems incurred, the Accident Investigation Board recommended that a four-line release be incorporated into the SR-71 personnel parachute to enhance control of these chutes and reduce oscillations.
- 3. ADP was requested to accomplish an Engineering Study and they recommended a six-line release rather than a four-line release modification to the MC-1 parachute, because of its larger size.
- 4. The parachute tests were performed at the Naval Test Center, El Centro, California. They included whirl tower tests at 175, 225, and 250 knots in addition to nine live jumps in "shirt-sleeve" environment and nine live jumps in full pressure suits.
- 5. The Jump Tests revealed:
  - a. The modified MC-1 parachute is very stable.



### Approved For Release 2002/11/08: CIA-RDP75B00285R000100080012-0

- b. The six-line release provided a slight turning improvement but actuation was difficult and tiring. Turns of 360 required 25-40 seconds to complete. Line release also induced a three to four knot forward speed; the desirability of this feature would vary with jump conditions, such as a 10 to 15 knot wind.
- c. Turns with the seat kit and raft released were more difficult to accomplish. The suspended survival kits made damping of crewmember oscillation more difficult.
  - d. Pressure suits did not present a problem in descent activity.
- 7. The test results have indicated that there is some merit to the six-line release modification. There is considerable doubt, however, that the parachute control exercised by test personnel would be utilized by crewmembers in an emergency situation. Test jumps were made with much preparation and full awareness of descent conditions, factors not usually available to crewmembers. Test personnel accomplished a variety of control maneuvers not expected of crewmembers.
- 8. ASPO engineering feels that the presently designed chute provides the optimum chances for survival under all circumstances and does not feel the proposed modifications to resolve speculation on possible adverse conditions will enhance the effectiveness of the existing system. We concur with the analysis and advice of the El Centro Parachute Test Center personnel and recommend that the modification not be accomplished.

FOR THE COMMANDER

WILLIAM MARSCHER, Lt Col, USAF Chief, Service Engineering Branch

will- March

Maintenance Engineering Division

Copies to: 9SRW/9DCOS SMAMA DET 51/FT IG/DSOFB/Lt Col Marsh SC-2

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